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S-5656**AMENDMENTS TO THE CLAIMS:**

Please amend the claims as follows. This listing of claims will replace all prior listings.

1. (CURRENTLY AMENDED) A force generator comprising:  
a rotationally fixed first circular member defined about a first axis to define a first inner diameter circular path, said first circular member having a first radius;  
a second circular member defined about a second axis offset from said first axis to define a second radius, said second radius one-half the radius of said first radius, said second circular member movable about the circular path to simultaneously complete one revolution about said second axis and one orbit around said first axis;  
a crank which mounts said second circular member, said crank rotatable about said first axis; and  
a mass located at-adjacent a circumference of said second circular member movable about a two-cusp hypocycloid path to generate a vibratory inertial force, to minimize a vibratory force.
2. (ORIGINAL) The force generator as recited in claim 1, wherein said vibratory inertial force is a sinusoidal inertial force in a straight line.
3. (CANCELED)
4. (CURRENTLY AMENDED) The force generator as recited in claim 1, wherein said rotationally fixed first circular member comprises a ring gear.
5. (ORIGINAL) The force generator as recited in claim 1, wherein said second circular member comprises a planet gear.

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6. (CANCELED)

7. (CURRENTLY AMENDED) The force generator as recited in claim-6 1, further comprising a motor which drives said crank.

8. (CURRENTLY AMENDED) The force generator as recited in claim-6 1, further comprising an opposed circular counter member mounted to said crank about a third axis offset from said first axis.

9. (CURRENTLY AMENDED) The force generator as recited in claim-~~4~~ 8, wherein said opposed circular counter member comprises a planet gear in meshed engagement with said first circular member.

10-12. (CANCELLED)

13. (CURRENTLY AMENDED) A method of force generation for active vibration control in a rotary-wing aircraft comprising the steps of:

- (1) defining a circular path within a rotationally fixed first circular member about a first axis, the first axis transverse to an axis of rotation of a main rotor assembly of the rotary-wing aircraft;
- (2) defining a second circular member about a second axis offset from the first axis;
- (3) locating a mass at-adjacent a circumference of the second circular member movable about a two-cusp hypocycloid path; and
- (4) controlling movement of the second circular member about the circular path such that the second circular member simultaneously completes one revolution about the second axis and one orbit around said first axis to generate a vibratory inertial force to minimize a vibratory force from the main rotor assembly.

14-18. (CANCELLED)

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19. (CURRENTLY AMENDED) ~~A force generator~~ An active vibration control (AVC) system for a rotary-wing aircraft comprising:

a multitude of sensors located within a rotary-wing aircraft;

a controller in communication with said multitude of sensors;

a force generator comprising:

a rotationally fixed first circular member defined about a first axis to define a first inner diameter circular path, said first circular member having a first radius;

a second circular member defined about a second axis offset from said first axis to define a second radius, said second circular member movable about the circular path to simultaneously complete one revolution about said second axis and one orbit around said first axis;

a crank which mounts said second circular member, said crank rotatable about said first axis;

a mass located on said second circular member movable about a two-cusp hypocycloid path to generate a vibratory inertial force; and

a motor which drives said crank, said motor in communication with said controller to drive said crank to control a phase and magnitude of said vibratory inertial force in response to said processor.

20-22. (CANCELED)

23. (CURRENTLY AMENDED) The force generator as recited in claim-22\_1, further comprising a motor which drives said crank.

24. (CURRENTLY AMENDED) The force generator as recited in claim 23, further comprising a controller which drives said motor to control said vibratory inertial force in

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response to ~~sensed~~ a vibratory forces from a main rotor assembly to minimize the vibratory force.

25-26. (CANCELED)

27. (NEW) The force generator as recited in claim 8, wherein said opposed circular counter member comprises a counterweight.

28. (NEW) The force generator as recited in claim 19, wherein said force generator is located within a fuselage of the rotary-wing aircraft.

29. (NEW) A method as recited in claim 13, further comprising the step of:

(5) sensing a vibratory force from a main rotor assembly; and

(6) transmitting the vibratory inertia force of said step (4) through a fuselage to minimize the vibratory force sensed in said step (1).

30. (NEW) A method as recited in claim 13, wherein said step (4) further comprises the step of:

(a) continuously varying the vibratory inertial force to a changing dynamic characteristic of the rotary-wing aircraft to minimize the vibratory forces sensed in said step (1).